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Solution: Initial conditions: $P_1 = 7.5 \text{ MPa}$, 5008 C $h_1 = 3404.3 \text{ kJ/kg}$ $s_1 = 6.7598 \text{ kJ/kg K}$ (h_1 and s_1 from superheated steam tables) At the exit state, $P_2 = 0.545 \text{ MPa}$; and therefore the nozzle is convergent. State 2 is fixed by $P_2 = 5 \text{ MPa}$, $s_2 = 6.7598 \text{ kJ/kg K}$ $T_2 = 4358 \text{ K}$, $v_2 = 0.06152 \text{ m}^3/\text{kg}$, $h_2 = 3277 \dots$

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We usually assume a single axisymmetric S2 surface and several untwisted S1 surfaces. The S2 (hub to tip or throughflow) solution has become the “ backbone ” of turbomachinery design. Initially there was rivalry between the matrix-stream function method and the streamline curvature method of solving the equations.

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